

Document: XP_P

XP_PT_TN_N01_R02

Author:

Arion Zimmermann

Mission:

European Rover Challenge

Date:

24.02.2021

Technical Note

Title: Power supply specifications

Prepared by: Arion Zimmermann

Reviewed by: Jakub Kucka

Approved by: ...

Revision History

Revision	Author	Description	Date
R01	Arion Zimmermann	Document Baseline	24.02.2021
R02	Arion Zimmermann	Clarified some requirements	3.03.2021

Table of content

Abstract	2
High-level requirements	2
Electrical specifications	4
References	9
Glossary	9



Mission: European Rover Challenge Date: 24.02.2021

1. Abstract

EPFL Xplore's Rover has many subsystems that must be powered at very specific voltages. The power supply will have to provide 5 different voltage channels, namely 3.3V, 5V, 15V, 24V, 48V and take, as an input, from 21V to 30V.

2. High-level requirements

The power supply unit (PSU) has 5 output channels and 1 input channel. The state of the converters between input and output must be controllable by a power supply controller (PSC). This controller must also receive active feedback from every converter.

The PSU must accept an input voltage of 21V to 30V. The DC input originates from a battery pack but passes through a circuit breaker and an automotive relay before arriving at the power supply's input.

Inside the PSU, a battery charge estimator counts how much charge is left in the battery pack and reports it to the controller. This charge estimator must take into account the voltage-charge profile of the battery cells.

Further, an undervoltage protection mechanism prevents the DC/DC converters from starting switching in an undervoltage condition, which could eventually result in unstable behaviour.

The PSU has to convert its input to different voltage levels. First of all, a 3.3V converter is necessary to power the rover's Avionics. Then, the Raman laser must be powered by a 5V input and 15V must be supplied to the main computer. Further, the LiDAR will have to be supplied with 24V and finally, the ethernet switches require an input voltage of 48V.

Each of these converters must shutdown if their respective connected load exceeds the maximum load they are designed for. Their output power must be measured at any time and reported to a supervising controller. This supervising controller must be able to shutdown any of the converters at any time and expects to receive a "power good" signal from the converter when the output voltage is within 10% of nominal output voltage.

The PSU's controller (PSC) controls the state of the DC/DC converters and receives information from those, as previously mentioned. This information must be transmitted by WiFi to the control station in a 2.4Ghz band and to the avionics through an I2C bus.

The output connectors of the DC/DC converters must accept cables with diameters ranging from AWG12 and AWG22. Specifications about the connector to the avionics are specified in RF01.



Mission: European Rover Challenge Date: 24.02.2021

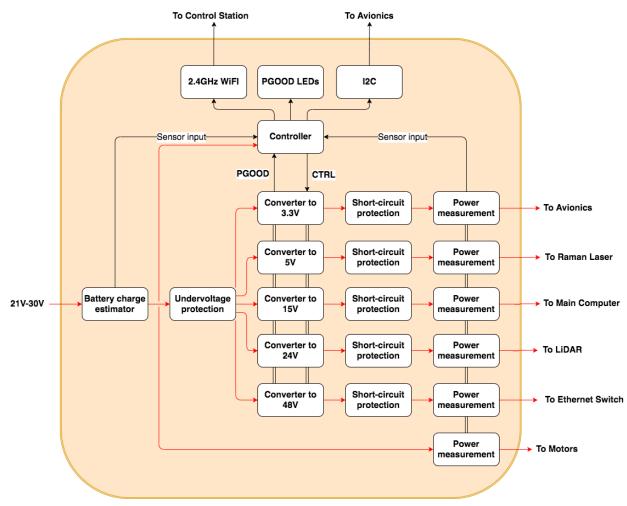


Fig. 1: Block diagram of the PSU architecture (in red: power path, in black: data path)



Mission: European Rover Challenge Date: 24.02.2021

3. Electrical specifications

Input characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Input voltage	Supplied from battery pack	21	25.4	29.4	٧
Source impedance	Measured at 1KHz	7	12	16	mΩ
Input current		-	-	25	Α

3.3V output characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Output voltage	At nominal load	3	3.3	3.6	٧
Output noise & ripple		-	50	100	mVpp
Output current		-	1.12	1.3	А
Output power		-	3.7	4.3	W
Nominal load		2.5	3	-	Ω
Efficiency	At nominal load	80	-	-	%
Efficiency	At half load	70	-	-	%

5V output characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Output voltage	At nominal load	4.8	5	5.2	V
Output noise & ripple		-	20	50	mVpp
Output current		-	4	5	Α
Output power		-	20	25	W
Nominal load		1	1.25	-	Ω
Efficiency	At nominal load	85	-	-	%
Efficiency	At halfload	80	-	-	%
Transient response	Response to ±50% step input	-	250	500	mV
Transient recovery	Response to ±50% step input	-	0.5	1	ms



Mission: European Rover Challenge Date: 24.02.2021

15V output characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Output voltage	At nominal load	9	15	19	V
Output noise & ripple		-	100	500	mVpp
Output current		-	2.2	3.2	Α
Output power		-	33.2	47.5	W
Nominal load		4.7	6.8	-	Ω
Efficiency	At nominal load	90	-	-	%
Efficiency	At half load	85	-	-	%

24V output characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Output voltage	At nominal load	22	24	26	V
Output noise & ripple		-	100	500	mVpp
Output current		-	0.75	0.85	Α
Output power		-	18	22	W
Nominal load		28	32	-	Ω
Efficiency	At nominal load	90	-	-	%
Efficiency	At half load	85	-	-	%
Transient response	Response to ±50% step input	-	1000	2000	mV
Transient recovery	Response to ±50% step input	-	1	5	ms

48V output characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Output voltage	At nominal load	44	48	57	V
Output noise & ripple		-	100	500	mVpp
Output current		-	0.81	0.87	А
Output power		-	39	41.5	W
Nominal load		55	59	-	Ω



Mission: European Rover Challenge Date: 24.02.2021

Efficiency	At nominal load	90	-	-	%
Efficiency	At half load	85	-	1	%
Transient response	Response to ±50% step input	-	2000	4000	mV
Transient recovery	Response to ±50% step input	-	1	5	ms



Mission: European Rover Challenge Date: 24.02.2021

Measurement characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Battery charge estimator precision	Integrator runs for at most one hour. Expected energy: 600Wh	-	2	5	%
Power measurement precision per channel		-	1	5	%
Power measurement CMRR per channel		80	100	-	dB

RF characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Operating frequency	Central frequency (excluding band channels)	-	2.4	1	GHz
RX sensitivity	Independant from antenna gain	-	-100	-90	dBm
TX power	Independant from antenna gain	18	19	-	dBm
TX signal power		-	-	1	W



Mission: European Rover Challenge Date: 24.02.2021

Mechanical characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
Length	From the battery pack's geometry	-	165	165	mm
Width	From the battery pack's geometry	-	120	120	mm
Height	From the battery pack's geometry	-	20	30	mm
Weight		-	300	400	g
PCB material	Specified by manufacturer (Aisler)	-	FR-4	-	-
Number of PCB layers	Specified by manufacturer (Aisler)	2	4	4	-

Thermal characteristics

Parameter	Comment	Min.	Тур.	Max.	Unit
PCB Temperature rise	Tested at full load (all outputs on) at 25°C	-	30	60	K



Mission: European Rover Challenge Date: 24.02.2021

4. References

Reference	Title	Description	Link
[RF01]		Specifies the requirements for the 5V channel.	Cobolt 08-01 Series
[RF02]		Specifies the requirements for the 3.3V channel. The avionics consists of 4 of those Nucleo STM32H745ZI-Q boards.	STM32H7 Nucleo-144 boards (MB1363) - User manual
[RF03]		Specifies the requirements for the 24V channel.	OS1 Mid-Range High-Resolution Imaging Lidar

5. Glossary

RF Reference

PSU Power supply unit

PSC Power supply control